

6.1R

6.1R (Acceleration) The velocity in a certain flow field is given by the equation

$$\mathbf{V} = 3yz^2\hat{i} + xz\hat{j} + y\hat{k}$$

Determine the expressions for the three rectangular components of acceleration.

(ANS: $3xz^3 + 6y^2z$; $3yz^3 + xy$; xz)

From expression for velocity, $u = 3yz^2$, $v = xz$, and $w = y$.

Since

$$a_x = \frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} + w \frac{\partial u}{\partial z}$$

then

$$\begin{aligned} a_x &= 0 + (3yz^2)(0) + (xz)(3z^2) + (y)(6yz) \\ &= \underline{3xz^3 + 6y^2z} \end{aligned}$$

Similarly,

$$a_y = \frac{\partial v}{\partial t} + u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} + w \frac{\partial v}{\partial z}$$

$$\begin{aligned} \text{and } a_y &= 0 + (3yz^2)(z) + (xz)(0) + (y)(x) \\ &= \underline{3yz^3 + xy} \end{aligned}$$

Also,

$$a_z = \frac{\partial w}{\partial t} + u \frac{\partial w}{\partial x} + v \frac{\partial w}{\partial y} + w \frac{\partial w}{\partial z}$$

$$\begin{aligned} \text{so that } a_z &= 0 + (3yz^2)(0) + (xz)(1) + (y)(0) \\ &= \underline{xz} \end{aligned}$$

6.2R

6.2R (Vorticity) Determine an expression for the vorticity of the flow field described by

$$\mathbf{V} = x^2y\hat{i} - xy^2\hat{j}$$

Is the flow irrotational?

(ANS: $-(x^2 + y^2)\hat{k}$; no)

The vorticity is twice the rotation vector :

$$\vec{\zeta} = 2\vec{\omega} = \nabla \times \vec{V} \quad (\text{Eq. 6.17})$$

From expression for velocity, $u = x^2y$, $v = -xy^2$, and $w = 0$, and with

$$\omega_x = \frac{1}{2} \left(\frac{\partial w}{\partial y} - \frac{\partial v}{\partial z} \right) \quad (\text{Eq. 6.13})$$

$$\omega_y = \frac{1}{2} \left(\frac{\partial u}{\partial z} - \frac{\partial w}{\partial x} \right) \quad (\text{Eq. 6.14})$$

$$\omega_z = \frac{1}{2} \left(\frac{\partial v}{\partial x} - \frac{\partial u}{\partial y} \right) \quad (\text{Eq. 6.12})$$

it follows that

$$\omega_x = 0, \quad \omega_y = 0, \quad \text{and} \quad \omega_z = \frac{1}{2} (-y^2 - x^2)$$

Thus,

$$\begin{aligned} \vec{\zeta} &= 2 (\omega_x \hat{i} + \omega_y \hat{j} + \omega_z \hat{k}) \\ &= 2 \left[(0) \hat{i} + (0) \hat{j} + \frac{1}{2} (-y^2 - x^2) \hat{k} \right] \\ &= \underline{\underline{- (x^2 + y^2) \hat{k}}} \end{aligned}$$

Since $\vec{\zeta}$ is not zero everywhere, the flow is not irrotational. No.

6.3R

6.3R (Conservation of mass) For a certain incompressible, two-dimensional flow field the velocity component in the y direction is given by the equation

$$v = x^2 + 2xy$$

Determine the velocity component in the x direction so that the continuity equation is satisfied.

(ANS: $-x^2 + f(y)$)

To satisfy the continuity equation,

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0 \quad (1)$$

Since $\frac{\partial v}{\partial y} = 2x$

Then from Eq. (1)

$$\frac{\partial u}{\partial x} = -2x \quad (2)$$

Equation (2) can be integrated with respect to x to obtain

$$\int du = -\int 2x dx + f(y)$$

or

$$u = \underline{\underline{-x^2 + f(y)}}$$

where $f(y)$ is an undetermined function of y .

6.4R

6.4R (Conservation of mass) For a certain incompressible flow field it is suggested that the velocity components are given by the equations

$$u = x^2y \quad v = 4y^3z \quad w = 2z$$

Is this a physically possible flow field? Explain.

(ANS: No)

Any physically possible incompressible flow field must satisfy conservation of mass as expressed by the relationship

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} = 0 \quad (1)$$

For the velocity distribution given,

$$\frac{\partial u}{\partial x} = 2xy, \quad \frac{\partial v}{\partial y} = 12y^2z, \text{ and } \frac{\partial w}{\partial z} = 2$$

Substitution into Eq. (1) shows that

$$2xy + 12y^2z + 2 \neq 0 \text{ for all } x, y, z.$$

Thus, this is not a physically possible flow field. No.

6.5R

6.5R (Stream function) The velocity potential for a certain flow field is

$$\phi = 4xy$$

Determine the corresponding stream function.

(ANS: $2(y^2 - x^2) + C$)

For the given velocity potential,

$$u = \frac{\partial \phi}{\partial x} = 4y \quad \text{and} \quad v = \frac{\partial \phi}{\partial y} = 4x$$

From the definition of the stream function,

$$u = \frac{\partial \psi}{\partial y} = 4y \quad (1)$$

Integrate Eq. (1) with respect to y to obtain

$$\int d\psi = \int 4y \, dy$$

or $\psi = 2y^2 + f_1(x)$ where $f_1(x)$ is an arbitrary function of x .

Similarly,

$$v = -\frac{\partial \psi}{\partial x} = 4x$$

and

$$\int d\psi = -\int 4x \, dx$$

or $\psi = -2x^2 + f_2(y)$ where $f_2(y)$ is an arbitrary function of y . (3)

To satisfy both Eqs. (2) and (3) $f_1(x) = f_2(y)$ for all x and y .

Thus, $f_1 = f_2 = \text{constant}$.

$$\psi = \underline{\underline{2(y^2 - x^2) + C}}$$

Where C is a constant.

6.6R

6.6R (Velocity potential) A two-dimensional flow field is formed by adding a source at the origin of the coordinate system to the velocity potential

$$\phi = r^2 \cos 2\theta$$

Locate any stagnation points in the upper half of the coordinate plane ($0 \leq \theta \leq \pi$).

(ANS: $\theta_s = \pi/2$; $r_s = (m/4\pi)^{1/2}$)

$$\phi = \frac{m}{2\pi} \ln r + r^2 \cos 2\theta, \text{ where } \phi_{\text{source}} = \frac{m}{2\pi} \ln r$$

Thus,
$$v_\theta = \frac{1}{r} \frac{\partial \phi}{\partial \theta} = -2r \sin 2\theta$$

and
$$v_r = \frac{\partial \phi}{\partial r} = \frac{m}{2\pi r} + 2r \cos 2\theta$$

Stagnation points will occur where $v_\theta = 0$, $v_r = 0$, for $0 \leq \theta \leq \pi$.

Thus,
$$0 = -2r_s \sin 2\theta_s \quad (1)$$

$$0 = \frac{m}{2\pi r_s} + 2r_s \cos 2\theta_s \quad (2)$$

Equation (1) is satisfied at $r_s = 0$ or $\theta_s = 0, \frac{\pi}{2}, \pi$.

From Eq. (2)
$$\cos 2\theta_s = -\frac{m}{4\pi r_s^2} \quad (3)$$

and for the possible values of θ_s , only $\theta_s = \frac{\pi}{2}$ will satisfy Eq. (3). Recall that $m > 0$ for a source.

Thus,
$$r_s = \sqrt{\frac{m}{4\pi}}$$

Thus, the stagnation point is located at

$$\underline{\underline{\theta_s = \frac{\pi}{2}}}, \quad \underline{\underline{r_s = \sqrt{\frac{m}{4\pi}}}}$$

6.7R (Potential flow) The stream function for a two-dimensional, incompressible flow field is given by the equation

$$\psi = 2x - 2y$$

where the stream function has the units of ft^2/s with x and y in feet. (a) Sketch the streamlines for this flow field. Indicate the direction of flow along the streamlines. (b) Is this an irrotational flow field? (c) Determine the acceleration of a fluid particle at the point $x = 1 \text{ ft}$, $y = 2 \text{ ft}$.

(ANS: yes; no acceleration)

- (a) Lines of constant ψ are streamlines. Thus, with $\psi = 2x - 2y$ the equation of a given streamline, ψ_1 , (where ψ_1 is some constant) is of the form

$$\psi_1 = 2x - 2y$$

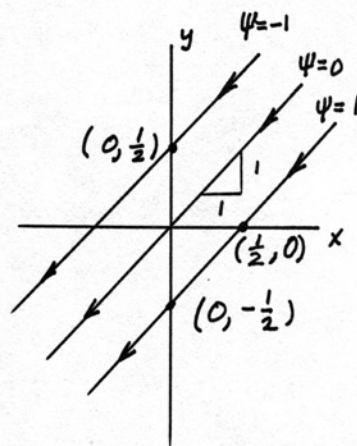
or

$$y = x - \frac{\psi_1}{2}$$

Thus, streamlines are straight lines as illustrated in the figure for three particular streamlines.

Since $u = \frac{\partial \psi}{\partial y} = -2$ $v = -\frac{\partial \psi}{\partial x} = -2$

the direction of flow is as shown on the figure



- (b) The flow field is irrotational if $\omega_z = 0$ where

$$\omega_z = \frac{1}{2} \left(\frac{\partial v}{\partial x} - \frac{\partial u}{\partial y} \right) \quad (\text{Eq. 6.12})$$

For the stream function given

$$\frac{\partial v}{\partial x} = 0 \quad \frac{\partial u}{\partial y} = 0$$

so that $\omega_z = 0$ and the flow field is irrotational. Yes.

- (c) Since the velocity is constant throughout the flow field, the acceleration of all fluid particles is zero.

$$\vec{a} = \frac{\partial \vec{V}}{\partial t} + \vec{V} \cdot \nabla \vec{V} = 0 \quad \text{since } \vec{V} = -2\hat{i} - 2\hat{j}$$

6.8R

6.8R (Inviscid flow) In a certain steady, incompressible, inviscid, two-dimensional flow field ($w = 0$, and all variables independent of z) the x component of velocity is given by the equation:

$$u = x^2 - y$$

Will the corresponding pressure gradient in the horizontal x direction be a function only of x , only of y , or of both x and y ? Justify your answer.

(ANS: only of x)

Since the flow field must satisfy the continuity equation

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

and with $u = x^2 - y$ it follows that

$$\frac{\partial v}{\partial y} = -\frac{\partial u}{\partial x} = -2x$$

and therefore

$$v = -2xy + f_1(x)$$

For steady, two-dimensional flow of an inviscid fluid (with the x -axis horizontal so that $g_x = 0$) the x -component of the momentum equation is

$$-\frac{\partial P}{\partial x} = \rho \left(u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} \right) \quad (\text{Eq. 6.51a})$$

Thus, for the u and v given above

$$\begin{aligned} \frac{\partial P}{\partial x} &= -\rho \left[(x^2 - y)(2x) + (-2xy + f_1(x))(-1) \right] \\ &= \rho \left[f_1(x) - 2x^3 \right] = F(x) \end{aligned}$$

The pressure gradient $\frac{\partial P}{\partial x}$ is a function only of x .